

ABSTRACT

A gas turbine engine with: a compressor section; a turbine section; a combustor, disposed between the compressor and turbine sections, having at least one combustor mounting assembly connecting the combustor to the engine. Each combustor mounting assembly has: a longitudinal axis; and an articulating joint having a first and second portion constrained from relative translation transverse to the longitudinal axis, and where said first and second portion have a multiple rotational degrees of freedom relative to each other about axes transverse to the longitudinal axis.